

# Influences of Variation in Geometry of Tape Insert on Thermal Performance and Flow Characteristics in Tubes

[Taiwo O. Oni and Manosh C. Paul]

**Abstract**—This paper presents the results of the numerical investigation carried out on different tube designs to assess the impacts of variation in the geometry of the twisted tape on heat transfer, friction factor and thermal performance of water inside the tubes. Alternate-axis triangular cut twisted tape was considered and its pitch and width, as well as the perimeter of the cuts on it were varied. The walls of the tubes were subjected to a uniform heat flux condition. The results of the investigation reveal that the thermal performance of the tube designs improves with an increase in the width of the tape and an increase in the size of the cuts on the tape but diminishes as the pitch of the tape increases.

**Keywords**—Twisted tape, heat transfer, thermal performance, designs

## I. Introduction

Due to the ease of installation, low cost and low maintenance, the use of twisted tape inserts has become a reliable technique for promotion of heat transfer [1, 2]. These techniques have found various applications in the process industries (air-cooled finned tube heat exchangers), automobile industries (radiators and internal heat exchangers) and residences (radiators, refrigeration systems and condensing central heating exchangers) [3-5]. The twisted tape generates a swirl flow which induces turbulence near the wall of the tube. This increase in the turbulent intensity near the tube improves fluid mixing in the tube and eventually leads to enhancement of heat transfer [2-6].

Some researchers have investigated the roles played by twisted tape in its application in heat transfer and they have made it known that the shapes of cuts made on the tape to modify them (the tape) has effects on the heat transfer promotion. Murugesan et al. [7] produced a trapezoidal-cut twisted-tape insert and later [8] fabricated a twisted-tape insert with square cuts. Chiu and Jang [9] reported on a

longitudinal strip insert with holes, Cui and Tian [10] fabricated an edge-fold twisted tape and Chang et al. [11] formulated a serrated twisted tape. These researchers as well as the findings from the other works indicated that the heat transfer produced by tubes induced with these modified twisted tapes is higher than those produced in the tubes without a twisted tape.

The present work is carrying out simulation on different tube designs with twisted tape having different geometries in their pitch, width and size of cuts made on them. The aim is to examine the effects which the variations of these geometries have on the heat transfer in the tubes.

## II. Geometry of Flow Models

Seven different flow models are considered in this study. Each of them consists of a tube induced with a twisted tape. The tube is 1000mm long and 19mm in diameter and the tape runs through the entire length of the tube. Each of the tapes has alternate axes and equilateral triangular cuts on them. The tape in each of the flow models is different from one another in width ( $w$ ), pitch ( $y$ ), length of a side of the triangular cut ( $s$ ) and perimeter of cut ( $p$ ) as demonstrated in Fig. 1 and Table I. The tapes are named as alternate-axis triangular cut twisted tape (ATCT) while the flow models are referred to as tube with alternate-axis triangular cut twisted tape (TATCT).

## III. Mathematical Formulation

The flow is assumed to be steady, and incompressible Newtonian and they are governed by the three-dimensional Navier-Stokes equations [12]. Turbulent flow is considered and the appropriate models are applied to determine the unknown variables in the RANS (Reynolds Averaged Navier-Stokes) equations as discussed below.

### A. Navier-Stokes Equations

The Navier-Stokes equations [21, 22] are given as

(a) Continuity equation:

$$\nabla \cdot \mathbf{V} = 0 \quad (1)$$

(b) Momentum equation:

$$\rho \frac{\partial \mathbf{V}}{\partial t} + \nabla \cdot (\rho \mathbf{V} \mathbf{V}) = -\nabla P + \nabla \cdot \mu (\nabla \mathbf{V}) \quad (2)$$

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This work enjoys sponsorship of Ekiti State University Ado-Ekiti Nigeria through a fund from the Tertiary Education Trust Fund Nigeria.

(c) Energy equation:

$$\rho C_p \left( \frac{\partial T}{\partial t} + \nabla \cdot \mathbf{VT} \right) = \nabla \cdot (k \cdot \nabla T) \quad (3)$$

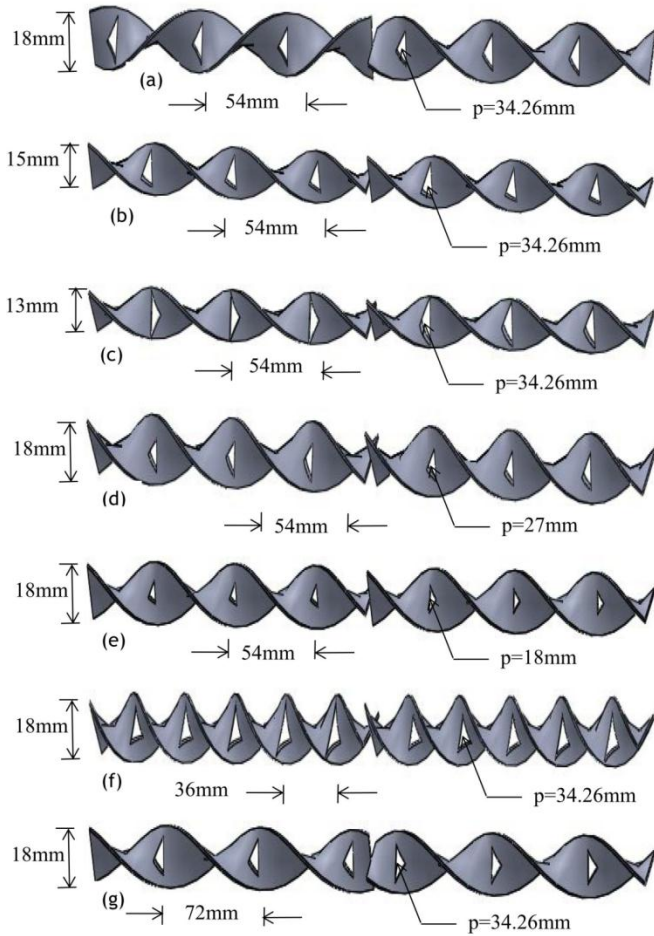


Figure 1. Geometries of the alternate-axis triangular cut twisted tape of (a) TATCT, (b) TATCTw1, (c) TATCTw2, (d) TATCTa1, (e) TATCTa2, (f) TATCTy1 and (g) TATCTy2.

TABLE I. DESCRIPTION OF THE VARIOUS GEOMETRIES OF THE ALTERNATE-AXIS TRIANGULAR CUT TWISTED TAPE (ATCT) USED FOR THE DOMAINS OF THE TATCT

Tape	Geometry			
	w (mm)	y (mm)	s(mm)	p (mm)
ATCT	18	54	11.42	34.26
ATCTw1	15	54	11.42	34.26
ATCTw2	13	54	11.42	34.26
ATCTa1	18	54	9.00	27.00
ATCTa2	18	54	6.00	18.00
ATCTy1	18	36	11.42	34.26
ATCTy2	18	72	11.42	34.26

where  $T$  is the temperature,  $\mu$  is the dynamic viscosity,  $P$  is the pressure,  $k$  is the thermal conductivity,  $\rho$  is the density and  $C_p$  is the heat capacity at constant pressure.

B.

### C. RANS Equations

The RANS equations [13, 14] are written as

(a) Mass conservation:

$$\frac{\partial}{\partial x_i} (u_i) = 0 \quad (4)$$

(b) Momentum equation:

$$\rho \frac{D\bar{u}_i}{Dt} = S_i - \frac{\partial \bar{P}}{\partial x_i} + \frac{\partial}{\partial x_j} \left( \mu \frac{\partial u_i}{\partial x_j} - \rho \overline{u'_i u'_j} \right) \quad (5)$$

(c) Time-averaged energy equation:

$$\frac{\partial}{\partial x_j} \rho C_p (u_j T) = u_j \frac{\partial P}{\partial x_j} + u'_j \frac{\partial P'}{\partial x_j} + \frac{\partial}{\partial x_j} \left( k \frac{\partial T}{\partial x_j} - \rho C_p \overline{u'_j T'} \right) \quad (6)$$

## iv. Turbulence Modelling

The turbulence models used for the simulation are discussed below.

### A. Standard $\kappa - \epsilon$ Model

The standard  $\kappa - \epsilon$  model [15] is based on turbulent kinetic energy ( $\kappa$ ) and dissipation rate of turbulence kinetic energy ( $\epsilon$ ). The transport equations [14] are given as:

$$\frac{\partial}{\partial t} (\rho \kappa) + \frac{\partial}{\partial x_i} (\rho \kappa u_i) = \frac{\partial}{\partial x_j} \left[ \left( \mu + \frac{\mu_t}{\sigma_\kappa} \right) \frac{\partial \kappa}{\partial x_j} \right] + G_\kappa - \rho \epsilon - Y_M + S_\kappa \quad (7)$$

$$\frac{\partial}{\partial t} (\rho \epsilon) + \frac{\partial}{\partial x_i} (\rho \epsilon u_i) = \frac{\partial}{\partial x_j} \left[ \left( \mu + \frac{\mu_t}{\sigma_\epsilon} \right) \frac{\partial \epsilon}{\partial x_j} \right] + C_{1\epsilon} \frac{\epsilon}{\kappa} G_\kappa - C_{2\epsilon} \rho \frac{\epsilon^2}{\kappa} + S_\epsilon \quad (8)$$

The turbulent viscosity ( $\mu_t$ ) is modelled as

$$\mu_t = \rho C_\mu \frac{\kappa^2}{\epsilon} \quad (9)$$

where  $G_\kappa$  is the generation of  $\kappa$  due to the mean velocity gradients;  $C_{1\epsilon}$ ,  $C_{2\epsilon}$ ,  $C_{3\epsilon}$  are model constants;  $\sigma_\kappa$  and  $\sigma_\epsilon$  are the turbulent Prandtl numbers for  $\kappa$  and  $\epsilon$  respectively;  $S_\kappa$  and  $S_\epsilon$  are user-defined source terms. The model constant are  $C_{1\epsilon} = 1.44$ ,  $C_{2\epsilon} = 1.92$ ,  $C_\mu = 0.09$ ,  $\sigma_\kappa = 1.0$ ,  $\sigma_\epsilon = 1.3$  [14].

## B. RNG $\kappa - \varepsilon$ Model

The Renormalization Group theory (RNG)  $\kappa$ - $\varepsilon$  model [16] has model transport equations for  $\kappa$  and  $\varepsilon$ . Its transport equations [14] are

$$\frac{\partial}{\partial t}(\rho\kappa) + \frac{\partial}{\partial x_i}(\rho\kappa u_i) = \frac{\partial}{\partial x_j} \left[ \alpha_\kappa \mu_{eff} \frac{\partial \kappa}{\partial x_j} \right] + G_\kappa + G_b - \rho\varepsilon - Y_M + S_\kappa \quad (10)$$

$$\frac{\partial}{\partial t}(\rho\varepsilon) + \frac{\partial}{\partial x_i}(\rho\varepsilon u_i) = \frac{\partial}{\partial x_j} \left[ \alpha_\varepsilon \mu_{eff} \frac{\partial \varepsilon}{\partial x_j} \right] + C_{1\varepsilon} \frac{\varepsilon}{\kappa} (G_\kappa + C_{3\varepsilon} G_b) - C_{2\varepsilon} \rho \frac{\varepsilon^2}{\kappa} - R_\varepsilon + S_\varepsilon \quad (11)$$

where  $\alpha_\kappa$  and  $\alpha_\varepsilon$  are the inverse effective Prandtl numbers for  $\kappa$  and  $\varepsilon$  respectively;  $C_{1\varepsilon} = 1.42$ ,  $C_{2\varepsilon} = 1.68$ ,  $C_\mu = 0.0845$ ,  $\sigma_\kappa = 0.7194$  and  $\sigma_\varepsilon = 0.7194$  are model constants [14].

## v. Boundary Conditions

Turbulent flow with Reynolds number between 5000 and 20000 were considered. A uniform heat flux condition is applied to the surface of tube wall and it is subjected to a no slip condition, i.e.  $u_i = u_j = u_k = 0$ . A velocity and temperature of 301K are specified at the inlet of the tube. The turbulence intensity,  $I$  is given by the expression in equation (12). At the outlet, the gauge pressure is set to zero, a backflow temperature is specified.

$$I = 0.16Re^{-0.125} \quad (12)$$

where  $Re$  = Reynolds number and  $D$  = tube diameter.

## vi. Numerical Technique

The governing partial differential equations mentioned above are solved by finite volume method. The calculation of pressure and velocity was coupled by the Semi Implicit Pressure Linked Equations (SIMPLE) algorithm [13]. Fluent [14] was used to obtain the iterative solution of the equations. Grid independence tests were conducted on the domains. As provided in Table II, the domains TATCT, TATCT<sub>w1</sub>, TATCT<sub>w2</sub>, TATCT<sub>a1</sub>, TATCT<sub>a2</sub>, TATCT<sub>y1</sub>,

TABLE II. GRID INDEPENDENCE TEST FOR VARIOUS DOMAINS OF TATCT

Domain	Cell		
TATCT	1658638	2515756	3018907
TATCT <sub>w1</sub>	947018	1042326	1904321
TATCT <sub>w2</sub>	810424	939615	1045580
TATCT <sub>a1</sub>	944241	1154115	1301128
TATCT <sub>a2</sub>	879649	1038263	1271045
TATCT <sub>y1</sub>	1002701	1247268	1381714
TATCT <sub>y2</sub>	1106413	1349955	1484286

TATCT<sub>y2</sub> become grid independent at cells 2515756,1042326, 939615, 1154115, 1038263, 1247268 and 1349955 respectively.

## vii. Results and discussions

The computational results are discussed in this section.

### A. Contour of Temperature

The temperature contour for the different tube designs can be found in Fig. 2. Frames (a), (b) and (c) show the tubes with tape width  $w = 18$ mm, 15mm and 13mm respectively. The tube with  $w = 18$ mm provides better temperature distribution than that with  $w=15$ mm and 13 mm. This is as a result of more efficient mixing inside the domain. As the perimeter ( $p$ ) of the cut is reduced from 43.26mm (frame a) to 27mm (frame d) and 18mm (frame e), the thermal boundary layer grows thinner and the temperature distribution is affected. When the pitch ( $y$ ) is reduced from 54mm (frame a) to 36 mm (frame f), the temperature distribution improves but reverse is the case when it is increased to 72mm (frame g).

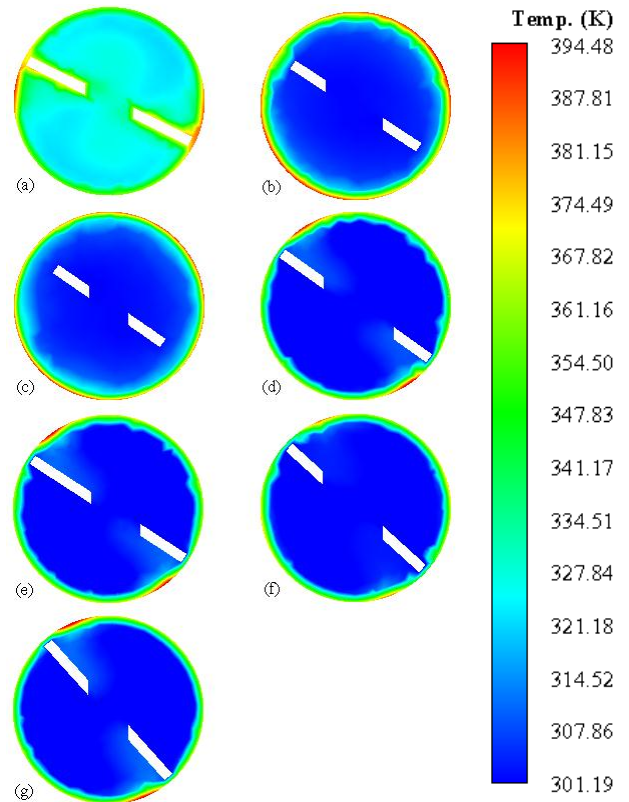


Figure 2. Contour plots of static temperature (K) for the domains of (a) TATCT, (b) TATCT<sub>w1</sub>, (c) TATCT<sub>w2</sub>, (d) TATCT<sub>a1</sub>, (e) TATCT<sub>a2</sub>, (f) TATCT<sub>y1</sub> and (g) TATCT<sub>y2</sub> for  $Re = 20000$ .

### B. Impacts of the Variations in the Tape Geometries on Heat Transfer

The heat transfer is expressed in term of Nusselt number. The effects of the variations in the geometries of the tape on the heat transfer are presented Fig. 3. The heat transfer decreases with decreasing width. This is because the domain with the largest free space (TATCT<sub>w2</sub>) between the tube wall and edge of the tape generates the weakest swirl flow, but the domain with the highest tape width (TATCT) creates the strongest swirl, resulting in effective transfer of heat across the layers of the fluids. The Nusselt number with tape geometry with  $w = 15\text{mm}$  (TATCT<sub>w1</sub>) and  $13\text{mm}$  (TATCT<sub>w2</sub>) are 6.1% and 8.9% respectively lower than that with  $w = 18\text{mm}$  (TATCT). The Nusselt number decreases with the decrease in the perimeter of cuts on the tape. The tape with smallest perimeter of cut therefore imparts lowest disturbance to the flow between the wall tube and edge of the tapes and correspondingly gives the lowest Nusselt number. The Nusselt number for the tape with  $p = 27\text{mm}$  (TATCT<sub>a1</sub>) and  $18\text{mm}$  (TATCT<sub>a1</sub>) are 2.5% and 7.9% respectively lower than that for the tape with  $p = 34.26\text{mm}$  (TATCT). When the pitch ( $y$ ) decreases from  $54\text{mm}$  to  $36\text{mm}$  the Nusselt number increases but decreases when the pitch increases from  $54\text{mm}$  to  $72\text{mm}$ . The tape with  $y = 72\text{mm}$  (TATCT<sub>y2</sub>) suffers a reduction in heat transfer rate of 3.4% of the tape with  $y = 54\text{mm}$  but the tape with  $y = 36\text{mm}$  (TATCT) enhances heat transfer rate up to 2.1% over that of the tape with  $y = 54\text{mm}$  (TATCT<sub>y1</sub>).

### C. Impacts of the Variations in the Tape Geometries on Friction Factor

It can be seen in Fig. 4 that as the tape width decreases from  $w = 18\text{mm}$  (TATCT) to  $15\text{mm}$  (TATCT<sub>w1</sub>) and then to  $13\text{mm}$  (TATCT<sub>w1</sub>) the friction factor also decreases. The value of the friction factor when  $w = 18\text{mm}$  decreases by 5.9% as compared with that when  $w = 15\text{mm}$  but decreases by 8.6% when  $w = 13\text{mm}$ . The friction factor decreases as the perimeter of the cut decreases. This is because the additional dissipation of pressure of the fluid caused by the fluid disturbance due to the presence of cuts on the tapes

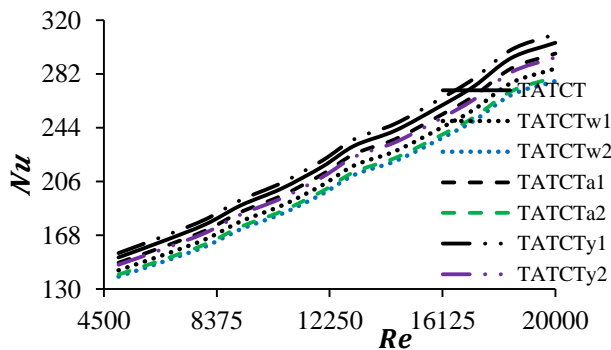


Figure 3. Effect of variations in tape geometries of TATCT on Nusselt number vs. Reynolds number.

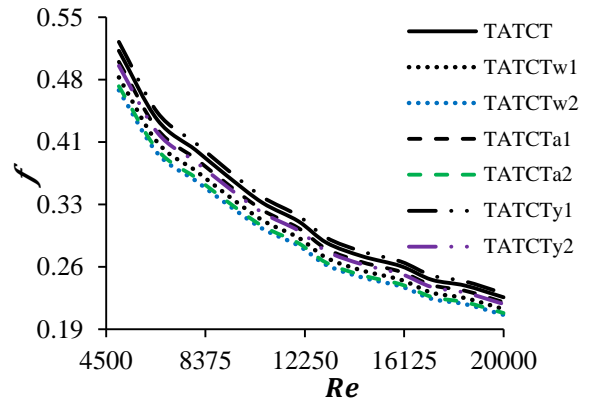


Figure 4. Effect of variations in tape geometries of TATCT on friction factor vs. Reynolds number.

resulted in an increase of interaction of the pressure force around a velocity boundary layer. The friction factors of the tape with  $p = 27\text{mm}$  (TATCT<sub>a1</sub>) and  $18\text{mm}$  (TATCT<sub>a2</sub>) are found to be lower than that of the tape with  $p = 34.26\text{mm}$  (TATCT) by 2.3% and 7.5% respectively. It is increases with the pitch decreasing. The friction factors obtained with the tape with pitch  $y = 36\text{mm}$  (TATCT<sub>y1</sub>) and  $72\text{mm}$  (TATCT<sub>y2</sub>) are 2.2% higher and 3.2% lower respectively than that obtained in the tape with  $y = 54\text{mm}$  (TATCT).

### D. Impacts of the Variations in the Tape Geometries on Thermal Performance Factor

The possibility of a twisted tape in promoting heat transfer is measured by thermal performance factor ( $\eta$ ) [8, 10], which is expressed as

$$\eta = \frac{Nu/Nu_p}{(f/f_p)^{1/3}} \quad (13)$$

where  $Nu_p$  and  $f_p$  are the Nusselt number and friction factor respectively of plain tube.

As demonstrated in Fig. 5, the thermal performance factor increases as the tape width and perimeter of cuts on the tape increase but decreases as the pitch of the tape increases. The performances got from the tape with  $w = 15\text{mm}$  (TATCT<sub>w1</sub>) and  $13\text{mm}$  (TATCT<sub>w2</sub>) are 1.4% and 3.3% respectively lower than that of the tape with  $w = 18\text{mm}$  (TATCT). With the tape with  $p = 27\text{mm}$  (TATCT<sub>a1</sub>) and  $18\text{mm}$  (TATCT<sub>a2</sub>) the performance factors are 1.1% and 3.1% respectively lower than that with  $p = 34.26\text{mm}$  (TATCT). The performance in TATCT<sub>y1</sub> ( $y = 36\text{mm}$ ) is enhanced up to 1.1% over that of the tape with  $y = 54\text{mm}$  (TATCT) but with the tape with  $y = 72\text{mm}$  (TATCT<sub>y2</sub>) the thermal performance diminished by 1.3% compared with that of the tape with  $y = 54\text{mm}$ .



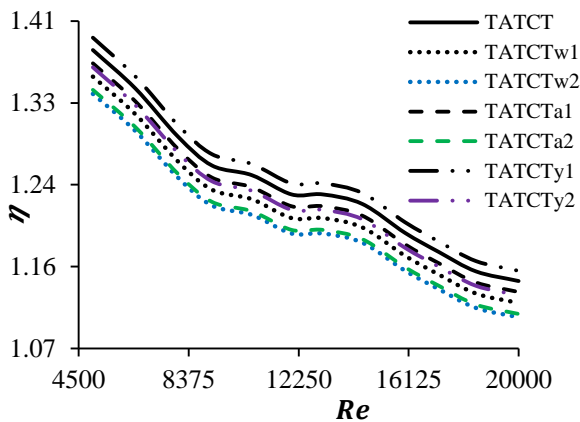


Figure 5. Effect of variations in tape geometries of TATCT on thermal performance factor vs. Reynolds number.

## VIII. Conclusion

In this study, the effects of variation in tape insert on heat transfer and thermal performance characteristics in a tube flow are investigated. It was discovered that increasing the tape width and the size of the cuts on the tape is favourable to the flow system as this augments the heat transfer and thermal performance of the system. Increasing the pitch of the tape resulted in the reduction of the performance of the system.

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